

Ship-Scale CFD Benchmark Study of a Pre-Swirl Duct on KVLCC2

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Abstract

Installing an energy saving device such as a pre-swirl duct (PSD) is a major investment for a ship owner and prior to an order a reliable prediction of

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the energy savings is required. Currently there is no standard for how such a prediction is to be carried out, possible alternatives are both model-scale tests in towing tanks with associated scaling procedures, as well as methods based on computational fluid dynamics (CFD). This paper summarizes a CFD benchmark study comparing industrial state-of-the-art ship-scale CFD predictions of the power reduction through installation of a PSD, where the objective was to both obtain an indication on the reliability in this kind of prediction and to gain insight into how the computational procedure affects the results. It is a blind study, the KVLCC2, which the PSD is mounted on, has never been built and hence there is no ship-scale data available. The 10 participants conducted in total 22 different predictions of the power reduction with respect to a baseline case without PSD. The predicted power reductions are both positive and negative, on average 0.4%, with a standard deviation of 1.6%-units, when not considering two predictions based on model-scale CFD and two outliers associated with large uncertainties in the results. Among the variations present in computational procedure, two were found to significantly influence the predictions. First, a geometrically resolved propeller model applying sliding mesh interfaces is in average predicting a higher power reduction with the PSD compared to simplified propeller models. The second factor with notable influence on the power reduction prediction is the wake field prediction, which, besides numerical configuration, is affected by how hull roughness is considered.

Keywords: Ship-scale CFD, Benchmark study, Pre-Swirl Duct, KVLCC2

1. Introduction

The strive towards more fuel efficient ships is motivated by both economic and regulatory reasons. The regulatory drive stems from the target of International Maritime Organization (IMO) to reduce the total annual greenhouse gas emissions from international shipping by at least 50% by 2050 compared to 2008 [1]. A possible approach to improve the energy efficiency of a ship is to equip it with an Energy Saving Device (ESD).

The construction and function of an ESD varies widely, from those mainly aiming to reduce the hull drag to others mostly focusing on the propulsive efficiency [2]. Other approaches, such as sails or air lubrication, are also possible alternatives [3]. The ESD in focus in this study may be referred to as a pre-swirl duct (PSD), which aims to improve the propulsive efficiency through modification of the axial and tangential velocity components of the propeller inflow, at the same time as it is generating thrust and unloads the propeller. Several studies focusing on the performance of various PSDs are available in the literature [4; 5; 6; 7].

Installing a PSD is a major investment for the ship owner and prior to an order a reliable prediction of the energy savings is required. For a PSD and similar propulsion improving devices there is currently no standard for how such prediction shall be carried out and how the expected power reduction should be reported. Possible alternatives are both model-scale tests in towing tanks with associated scaling procedures, as well as different methods based on computational fluid dynamics (CFD).

Model-scale tests, and model-scale CFD, are limited by their inability to match the Reynolds number of the ship, which is practically impossible as

26 Froude number similarity is normally required. This implies thicker bound-
27 ary layers and a larger wake on the model in relation to that on the ship.
28 The thicker boundary layers on the model are more prone to separation. Ad-
29 ditionally, the low Reynolds number may result in laminar boundary layers
30 on parts of the propulsion system. A PSD is operating in the wake, and
31 hence the differences between the model and ship wake field are critical for
32 its performance. However, still several CFD-studies of energy saving devices
33 are focusing on model-scale performance, as for instance [8], due to the avail-
34 ability of experimental data and possibility of local flow validations. Song
35 et al. [9] shows for a specific ESD that its potential gain could be of the
36 same order of magnitude as the uncertainties caused by scale effects, which
37 clearly indicates its unsuitability for predictive purposes. To account for the
38 scale-effects the International Towing Tank Conference (ITTC) has proposed
39 specific scaling procedures. In 1999 they suggested a method where the PSD
40 should be considered as a part of the hull in the resistance and self-propulsion
41 tests combined with a modified scaling of the wake fraction [10]; the method
42 was however never added to the ITTC recommended procedures and guide-
43 lines [2]. Recently, a different method has been suggested by ITTC, with
44 another approach to obtain the ship-scale wake fraction as well as thrust
45 deduction factor [2]. The general validity of both methods is however still
46 an open question. Another approach is suggested for scaling of the Mewis
47 duct, where the power saving observed for the ship is assumed to be very
48 similar to that measured in model-scale [5]. However, this holds under the
49 condition that the Mewis duct geometry is adjusted to the ship-scale flow
50 based on differences between model and ship-scale CFD. One more possible

51 method to obtain a prediction of the power reduction is to construct a wake
52 field in model-scale that more resembles the ship wake field, as for instance
53 conducted in [4].

54 An alternative to model-scale testing and associated scaling procedures,
55 to avoid the influences from Reynolds number differences between model
56 and ship, is the use of ship-scale CFD. The current status of ship-scale CFD
57 for power prediction is reviewed by Terziev et al. [11], where the principal
58 bottlenecks for replacing testing and extrapolation with ship-scale CFD are
59 identified to be the availability of open full-scale data, including ship geome-
60 tries, and computational power to predict the flow with sufficient accuracy.
61 Model-scale CFD has reached a relatively high level of maturity through the
62 international workshop series on CFD in Ship Hydrodynamics, held since
63 1980 [12]. Currently, the only to some extent corresponding workshop for
64 ship-scale CFD is the Lloyds Register workshop held in 2016 [13]. This
65 workshop only included overall values to use for validation, not any detailed
66 flow measurements. However, flow measurements on ships including ship
67 geometries, possible to use for CFD validation, are available in partly con-
68 fidential data sets, as reported by for instance Inukai et al. [14], Sakamoto
69 et al. [15], and Wakabayashi et al. [16]. There is also an industry wide
70 research project ongoing to provide more ship-scale data possible to apply
71 for CFD validation [17], neither this data set is presently openly available.
72 However, the next occasion of the international workshop series on CFD in
73 Ship Hydrodynamics [12], is planned to include a ship-scale validation case
74 for the first time.

75 Despite the general lack of detailed validation data, several ship-scale

76 CFD studies have been conducted and published. Pereira et al. [18] con-
77 ducted computations on KVLCC2 in both model and ship-scale and con-
78 cluded that the scale-effects are larger than the numerical uncertainties and
79 also that the wake-fraction reduction from model to ship-scale is clearly de-
80 pendent on the selected turbulence model. Orych et al. [19] had access to
81 both experimental data and ship trial results (confidential data) for a cargo
82 vessel and conducted a CFD validation and verification exercise with the
83 conclusion that there were no significant differences in uncertainty levels be-
84 tween model and ship-scale computations. Another study by Sun et al. [20]
85 comparing CFD-predictions with sea trial results claimed that various free
86 surface treatments contributed with up to 5 % uncertainty in power pre-
87 diction, and that roughness could have an up to 7% effect on the delivered
88 power. Similarly Niklas and Pruszko [21] concluded that their ship-scale CFD
89 results varied from -10 % to +4 % in relation to sea trials data, dependent on
90 hull roughness assumption and turbulence model. The flow measurements
91 and calculations by Sakamoto et al. [15] indicated the necessity to account
92 for hull roughness modelling. However, the scarce amount of data covering
93 both flow measurements and detailed hull surface characterizations, implies
94 that hull roughness modelling, in combination with near wall modelling and
95 turbulence modelling, is an aspect currently associated with high levels of
96 uncertainty.

97 The research area of ship-scale CFD is thus slowly evolving, but with
98 disparate conclusions on how to prioritize the efforts and the reliability of
99 any single computation. However, ship-owners have an urgent need for more
100 reliable ESD energy saving predictions to be able to oblige to current and up-

101 coming regulations. This urge motivates this study for which the objective
102 is to compare industrial state-of-the-art ship-scale CFD predictions of the
103 power reduction through a PSD installation on KVLCC2. The comparisons
104 will focus on how various CFD modelling aspects influence the power reduc-
105 tion prediction, which hopefully can be an aid for further ship-scale CFD
106 development and validation work, indicating where efforts on improving pre-
107 dictions should be focused, as well as a useful reference for ship-owners when
108 deciding upon a possible ESD installation based on ship-scale CFD predic-
109 tions. The variations in computational configurations among the benchmark
110 submissions include, among others, choice of turbulence model, propeller
111 modelling approach, consideration of superstructure drag, and how to in-
112 clude hull roughness.

113 **2. Organization of Study**

114 This CFD benchmark study is organized by SSPA Sweden AB and Chalmers
115 University of Technology as part of a research project initiated due to the lack
116 of standard for prediction of expected power reduction for ESDs and how this
117 should be reported to a customer. All participants were invited under the
118 condition that they had to participate at their own expenses. Everyone was
119 supplied with the same instructions, a description of the operating conditions
120 and ship-scale geometries, as well as CAD-files. It was clearly stated that the
121 main aim of the study was to predict the power reduction through the PSD
122 installation applying CFD, i.e. with less focus on the absolute power predic-
123 tion. This implies that the minimum number of computations were two, self-
124 propulsion with and without the PSD mounted on the ship, but alternative

125 CFD setups were warmly welcomed. The details on the CFD setup, compu-
126 tational grids, results and a qualitative uncertainty self-assessment were to
127 be summarized in a provided Excel-template. The ship has never been built
128 and hence there is no ship-scale data available for validation. Model-scale
129 tests with and without the PSD have been conducted at SSPA, however these
130 results have not been disclosed to any participants before the submission of
131 the predictions. This article is accompanied by a publicly-available data set¹.
132 It includes instructions as provided in the CFD benchmark study, geometries
133 and a compilation of details (CFD setup, computational grids, and results)
134 for the submitted predictions.

135 **3. Description of Geometry and Operating Conditions**

136 The original KVLCC2 hull designed as a test case for CFD around 1997 is
137 selected, which is the one used and described for instance in the 2010 Work-
138 shop in Ship Hydrodynamics [22]. Some minor geometrical modifications are
139 introduced to the hull to obtain a watertight geometry for production pur-
140 poses at SSPA, which has resulted in small differences in wetted surface area
141 and displacement as well as LCB (longitudinal centre of buoyancy). The
142 main particulars of the hull are provided in Table 1. It is in this benchmark
143 study assumed that the hull is coated with a traditional anti-fouling paint
144 which is applied according to instructions from paint manufacturers and that
145 the measured Average Hull Roughness (AHR) can be assumed to 100 μm .
146 Further, the transverse projected area above the water of the ship including

¹https://figshare.com/projects/Ship-Scale_CFD_Benchmark_Study_of_a_Pre-Swirl_Duct_on_KVLCC2/133095

147 superstructures (A_T) is assumed to be 1200 m^2 in this study.

Table 1: Main particulars of KVLCC2.

Length between perpendiculars, L_{PP} [m]	320
Beam, B [m]	58.0
Draft, T [m]	20.8
Displacement, Δ [m^3]	312 784
Wetted surface area without rudder, S_W [m^2]	27 249
Wetted surface area of rudder, S_{WR} [m^2]	273.3
Block coefficient, C_b	0.8098
LCB (forward of $L_{PP}/2$)	3.499%

148 The propeller is the one applied for the model tests with and without
149 PSD at SSPA. It is similar to the one designed by MOERI, but not exactly
150 the same propeller as used in previous workshops. The main particulars of
151 the propeller are listed in Table 2. The propeller is longitudinally positioned
152 6.4 m from the aft perpendicular, and vertically 5.992 m above the baseline.
153 A side view of the hull with propeller and rudder is shown in Figure 1.



Figure 1: KVLCC2 hull with propeller and rudder.

154 The PSD is designed by SSPA exclusively for this study. It is equipped
155 with three stator blades as shown in Figure 2. The hull, propeller and rudder
156 are identical for the case with the PSD mounted and for the case without

Table 2: Main particulars of propeller (R = propeller radius, P = pitch).

Propeller diameter, D_P [m]	9.86
Hub diameter, D_H [m]	1.528
Number of blades	4
Expanded blade area ratio, EAR	0.426
Chord length at 0.7R [m]	2.305
P/D_P at 0.7R	0.721
Max camber at 0.7R [m]	0.0501

157 PSD.

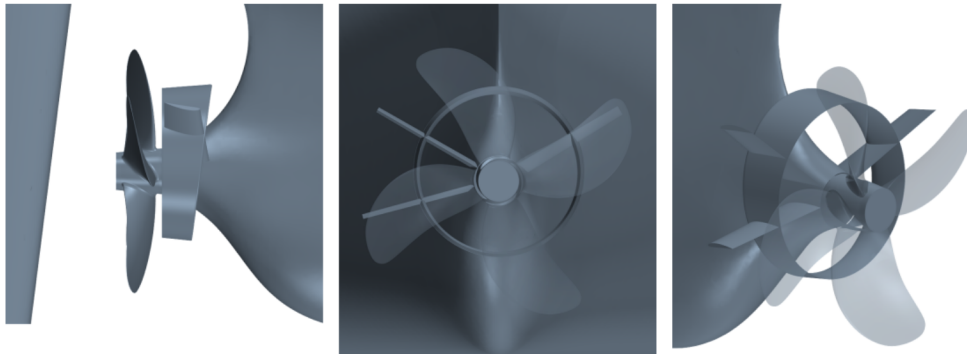


Figure 2: Pre-swirl duct (PSD) applied within this study.

158 The operating conditions are assumed to be optimal sea trial conditions,
 159 i.e. no currents, waves or wind to account for. The speed of the vessel applied
 160 in the CFD benchmark study is 15.0 knots, and not 15.5 knots as commonly
 161 used for KVLCC2 in workshops. 15 knots corresponds to a Reynolds number
 162 (Re) based on L_{PP} of $2.3 \cdot 10^9$ and Froude number (Fn) of 0.138. Salt water
 163 is assumed, with a water temperature of 20 °C, and air temperature 15 °C.

164 4. Summary of Submitted CFD-computations

165 In total 13 different organisations participated in the study, mainly uni-
166 versities and research institutes/ship-model basin organisations, but also one
167 organization related to a software supplier and one independent CFD con-
168 sultant firm. A few of them collaborated so the total number of partici-
169 pants should rather be counted as 10. The submitted results are obtained
170 through the use of eight different CFD software: STAR-CCM+, FreSCo+,
171 FINE/Marine, HELYX, OpenFOAM, NaViiX, NAGISA and SHIPFLOW.
172 In total 22 different predictions of the power reduction through a PSD in-
173 stallation on KVLCC2 are done.

174 All predictions are based on ship-scale CFD, except from two which are
175 model-scale CFD results extrapolated to ship-scale using scaling procedures.

176 4.1. CFD Setup

177 All submissions are based on the Reynolds-Averaged Navier-Stokes (RANS)
178 equations. Turbulence is modelled using a variety of one- and two-equation
179 turbulence models, namely: SST $k - \omega$, SST $k - \omega$ with QCR and curvature
180 correction, $k - \omega$ (Wilcox), $k - \varepsilon$, Spalart-Allmaras, LEASM $k - \omega$, EASM-
181 BSL with curvature correction, and EASM. The most frequently used model
182 is the ordinary SST $k - \omega$ model which is applied in 12 out of 22 submissions.

183 The free surface is discretized using the Volume-of-fluid (VOF) method in
184 seven submissions. For the remaining 15 submissions the setup is simplified
185 using a symmetry plane instead of the free surface, commonly referred to as
186 a double-body model.

187 Free sinkage and trim is allowed for in six of the submissions, while one

188 submission is based on sinkage and trim results obtained in a simplified setup.
189 The predicted sinkage and trim of the ship has a relatively low variation
190 between submissions, and is also similar between the PSD and reference
191 cases, for all submissions except from one outlier with significant difference
192 in sinkage and trim between the cases. Except for the outlier, the predicted
193 sinkage are all in the span 0.28-0.34 m and the trim predictions varies between
194 -0.09° and -0.125° (defined as positive when bow is up).

195 The detailed superstructure of the ship is not known, only the assumed
196 transverse projected area. 13 of the submissions account for air resistance
197 using a correlation which results in air resistance of 1.3-2.4% of the total
198 resistance of the ship, with a majority of the results within the upper range.
199 Eight of the submissions do not account for air resistance. One submission
200 tries to model the air resistance using a simplified superstructure, which
201 results in a lower resistance than that obtained using the correlations.

202 To account for the hull surface conditions as outlined in Section 3, hull
203 roughness is modelled in nine of the submissions using a variety of roughness
204 functions and equivalent sand grain roughness heights. Seven of the submis-
205 sions account for the additional resistance the roughness implies through a
206 correlation, but do not model it in CFD, hence no influence on the boundary
207 layers are accounted for. Six of the submissions do not account for the hull
208 roughness at all.

209 The propeller is geometrically represented using sliding mesh interfaces
210 in 10 of the submitted predictions. The other submissions are based on sim-
211 plified propeller models. The ones applied, as described by each participant,
212 are: a lifting line method, a hybrid lifting line/surface method (Yamazaki

213 model), a boundary element method (BEM), a body force model combined
214 with propeller open water curve, and a potential theory-based infinite-bladed
215 propeller model. In one submission the propeller is geometrically repre-
216 sented but with the motion modelled using a moving reference frame (MRF).
217 Amongst the submissions using sliding mesh interfaces it is most common
218 to obtain thrust-resistance balance through manual variation of the rotation
219 rate or to apply load variation (British method) and determine the operating
220 point by interpolation. For the simplified propeller models it is most common
221 with an automatic adjustment of the rotation rate.

222 *4.2. Computational Grid*

223 The grids are constructed using seven different software: STAR-CCM+,
224 HEXPRESS, helyxHexMesh, snappyHexMesh, Pointwise, UP-GRID, and
225 SHIPFLOW. The majority of the grids are unstructured, except for three
226 submissions which apply structured grids.

227 All submissions, except from two, apply wall functions, where one of the
228 submissions resolving the boundary layers is simulating the ship at model-
229 scale. The total number of cells and average y^+ on the hull below the water
230 surface is shown for all submissions in Figure 3. Both numbers varies widely
231 between the submissions. No obvious correlations of the total number of
232 cells to the free-surface modelling approach, propeller modelling approach
233 or boundary layer resolution are noted. Amongst the submissions based on
234 ship-scale CFD the mean number of cells is 19 million (median = 15 million)
235 and the median value for y^+ equals 200. With regards to the boundary layer
236 resolution, for the submissions based on ship-scale CFD, it is most common
237 to apply a total thickness of the prism layers on the hull of 0.3-0.5% of L_{PP} ,

238 two submission apply a lower value and three a higher (for the submissions
 239 applying structured grids this factor is not relevant). The applied expansion
 240 ratio between the prism layers in the boundary layer varies between 1.2 and
 241 1.5.

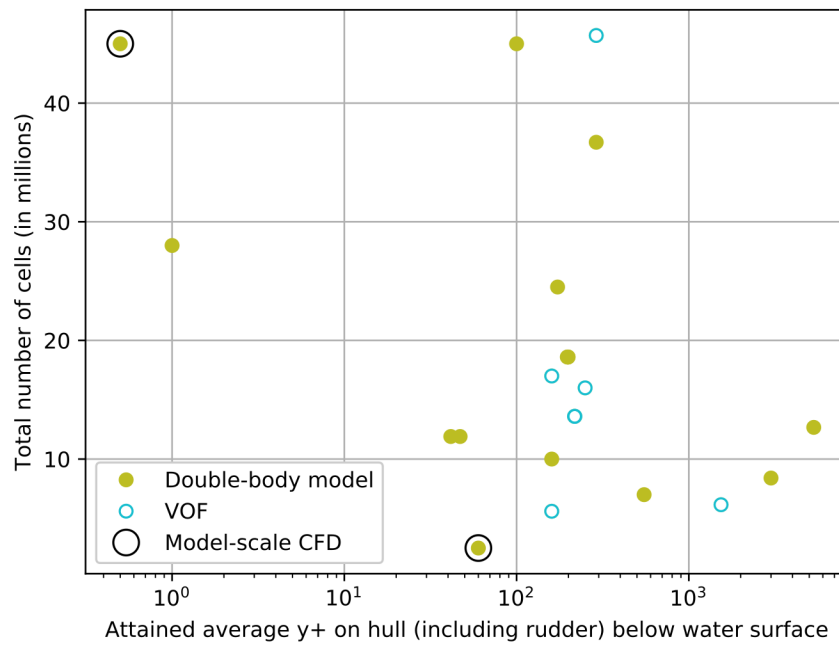


Figure 3: Total number of cells and average y^+ for all submissions.

242 The number of cells within the propeller domain and average y^+ on the
 243 propeller is shown for all submissions with a geometrically resolved propeller
 244 applying sliding mesh interfaces in Figure 4. A large variation is noted for
 245 both these variables amongst the submissions.

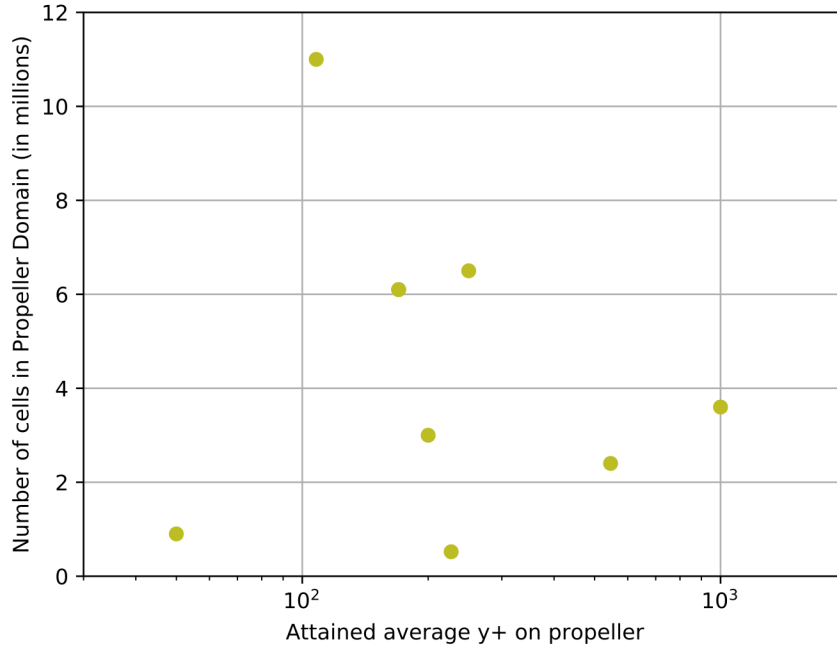


Figure 4: Total number of cells within the propeller domain and average y^+ on propeller for sliding mesh submissions.

246 *4.3. Computational Cost*

247 All participants have estimated the time for delivery of prognosis, counted
 248 from when the complete geometry and list of operating conditions are ob-
 249 tained. The estimations varies between two days and two months, but with
 250 the majority claiming approximately one week. The total required time for
 251 delivery of the prognosis is amongst other factors dependent on availability
 252 of computational resources and required number of core-hours. The total
 253 required core-hours, i.e. for both the reference case and the case with PSD,
 254 are shown in Figure 5. Except from two outliers the requirements varies

255 between 450 and 15 400 core-hours, with a median value of 4000 core-hours.
 256 The predictions based on sliding mesh generally requires more computational
 257 resources, however there is a large variation. This variation is partially re-
 258 lated to the required number of propeller revolutions at the final rotation
 259 rate, which varies between 5 and 600, with a median value of 24 revolutions.

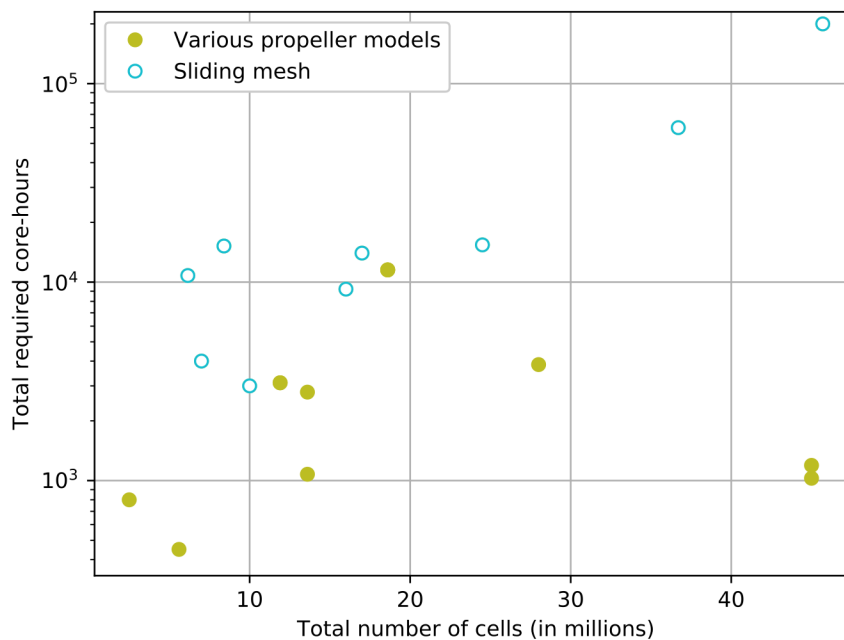


Figure 5: Total required number of core-hours in relation to the total number of cells for all submissions.

260 5. Comparison of Results

261 The main aim is to predict the power reduction through the PSD instal-
 262 lation, hence less effort could be dedicated to the absolute power prediction.

263 In Figure 6 the power difference between the case with PSD and the refer-
264 ence case is plotted against the predicted power for the reference case. For
265 each result it is indicated if the power prediction is considered as a repre-
266 sentative power prediction by the user, in other words if the setup had been
267 the same if the main aim was to predict the absolute power. The power
268 differences are presented in relative terms, and defined so that a negative
269 difference implies a power reduction with the PSD. Except from two outliers
270 at about -10% and +10%, the predicted power difference through installing
271 the PSD varies between -2.9% to +3.4%. The two outliers can be explained
272 by large differences in sinkage and trim, as described in Section 4, and dif-
273 ficulties in obtaining thrust-resistance equilibrium, respectively. Due to the
274 large uncertainties associated with these outliers, they will not be included
275 in the further analysis of the results. Neither will the two submissions based
276 on model-scale CFD be included in the remaining analyses, since the de-
277 tailed CFD results are not comparable and additional uncertainties due to
278 the scaling procedures are included. The average predicted power difference,
279 with the outliers and model-scale results excluded, is -0.4% with a standard
280 deviation of 1.6%-units.

281 Figure 6 illustrates also a relatively large spread in the predicted power,
282 even when only the predictions that are considered representative by each
283 user are taken into account. The mean predicted power amongst the predic-
284 tions considered representative is 17 724 kW, with a standard deviation of 2
285 026 kW, corresponding to 11% in relative terms.

286 That the PSD is not working properly, which is indicated by the predicted
287 power differences which varies around zero, could be partially explained by

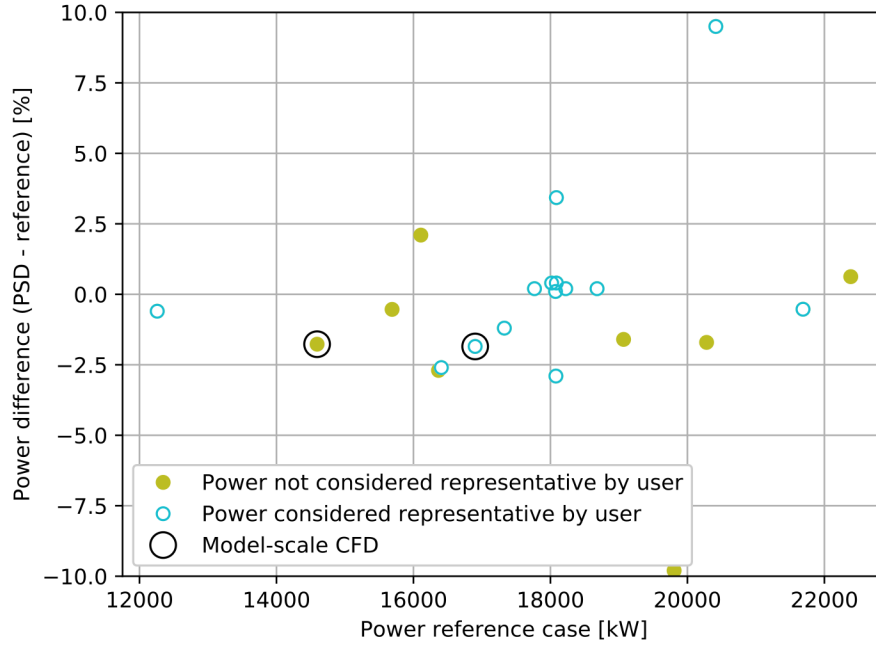


Figure 6: Power difference against absolute power for the reference case.

288 regions of unfavorable flow separation on the PSD indicating an unsatisfying
 289 alignment of the stators, as shown in Figure 7. Especially this is noted at
 290 the root of the stators.

291 Even if it is not the main objective of this study, it is interesting to
 292 compare the ship-scale CFD predictions with the model-scale test results
 293 from SSPA. Extrapolated to ship-scale, using the ITTC 1978 [23] and 1999
 294 methods [10], the model-scale tests predict a power reduction of 4.4%, i.e.
 295 significantly more than the ship-scale CFD predictions. Further, the model-
 296 scale tests predict the power for the reference case to 16 858 kW, i.e. within
 297 one standard deviation from the mean predicted power amongst the CFD

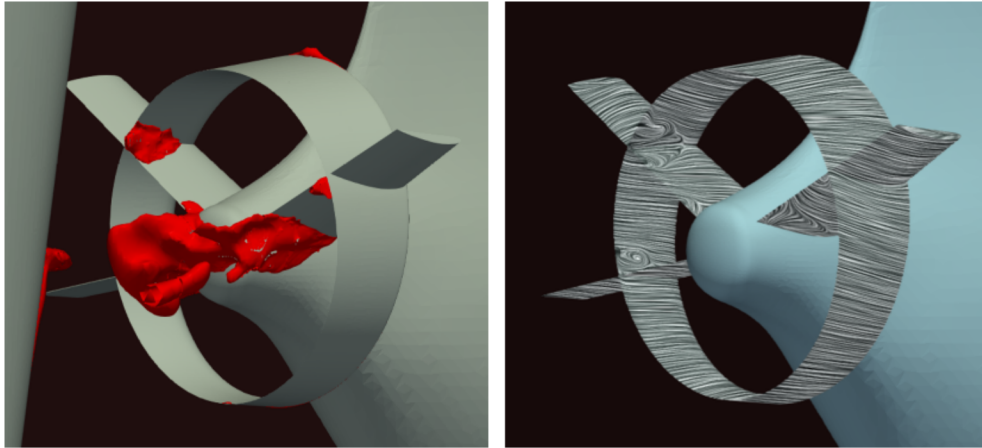


Figure 7: Iso-surface of negative axial velocity (left) and streamlines on PSD (right) illustrating the flow separation for a submission predicting a 0.2% power increment with the PSD.

298 submissions.

299 In Figure 8, the power difference is plotted against the propeller thrust
 300 difference for the data set without outliers and results based on model-scale
 301 CFD. There seems to be a correlation between the power difference and pro-
 302 peller thrust difference, which seems reasonable; a reduced propeller thrust
 303 implies that it is unloaded by the PSD which generates thrust. A reduced
 304 propeller thrust most probably also implies a reduced torque, which together
 305 with the rotation rate defines the power. On the other hand, when the PSD
 306 installation creates additional drag and the propeller needs to produce more
 307 thrust, an increment in power is noted if the increment in torque dominates
 308 over rotation rate differences. Additionally, Figure 9 shows the power differ-
 309 ence against the rotation rate difference, also for the data set without outliers
 310 and predictions based on model-scale CFD.

311 In Figures 8 and 9 the predictions obtained using a geometrically resolved

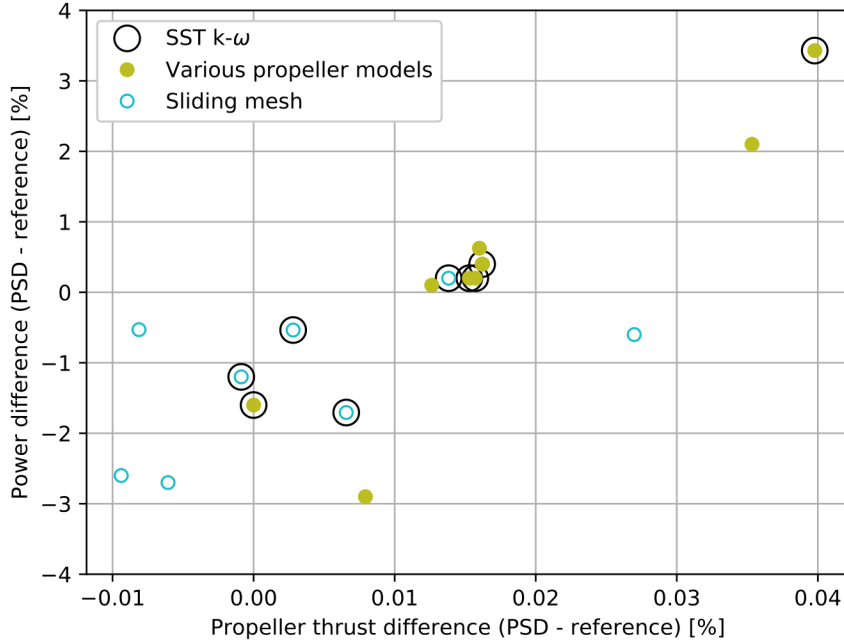


Figure 8: Power difference against propeller thrust difference.

312 propeller with sliding mesh interfaces are marked. There is an indication that
 313 the predictions applying a geometrically resolved propeller in general implies
 314 a better performance of the PSD. The average predicted power difference for
 315 the subset using sliding mesh interfaces is -1.2% with a standard deviation
 316 of 1.0%-units. For the predictions using various simplified propeller models
 317 the average predicted power difference is +0.3% with a standard deviation of
 318 1.6%-units. Since large differences between the simplified propeller models
 319 are expected, and due to the fact that the results are influenced by other
 320 modelling aspects, a simple explanation to this observation cannot be put
 321 forward and further studies are necessary. However, a possible theory may

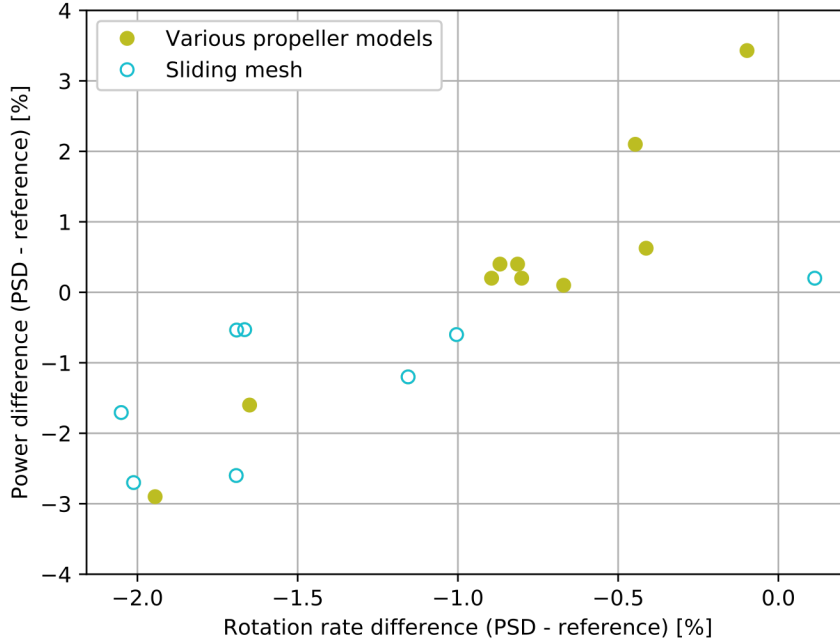


Figure 9: Power difference against rotation rate difference.

322 be related to the simplified propeller models representativeness at off-design
 323 conditions. Figure 10 illustrates the torque for one blade around a revo-
 324 lution as obtained in a submission predicting a power difference of -2.6 %.
 325 The PSD is redirecting the flow and increasing the blade torque, hence the
 326 angle of attack, especially when the blade is lightly loaded; the stators are
 327 located approximately at 60°, 255° and 300°. A simplified propeller model
 328 which is not fully representative at light load (i.e. high advance ratio), may
 329 over- or underestimate the efficiency of the propeller at these locations. If
 330 it would be so that the simplified propeller models applied in this study, to
 331 the largest extent are overestimating the propeller efficiency at light load,

332 the propeller will not suffer as much in the reference case as a resolved pro-
 333 peller would indicate. Hence, the gain of adding a PSD would be lower, or
 334 even negative, as noted from the results. Additionally, simplified propeller
 335 models may lack in their ability to resolve the temporal fluctuating flow be-
 336 hind the stators, which may impact the possibility to accurately represent
 337 the propulsion system performance with a PSD included. It is worth to note
 338 that in a comparison in model-scale with a pre-duct without stators on the
 339 JBC test case [12], a similar trend for geometrically resolved propellers versus
 340 simplified propeller models was not seen.

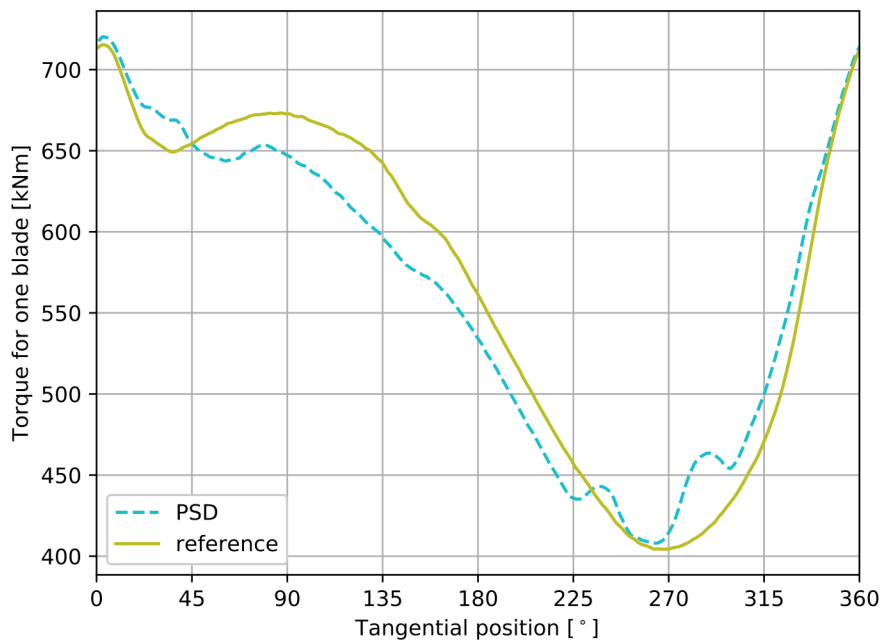


Figure 10: Torque for one blade around a revolution for one of the submissions.

341 Figure 8 also indicates the predictions applying the ordinary SST $k - \omega$

342 model, these results show a better correlation between thrust and power
 343 difference. The three main outliers in this plot are based on $k - \omega$ (Wilcox),
 344 EASM and Spalart-Allmaras. This indicates that the choice of turbulence
 345 model matters, even if there is no obvious correlation to the predicted power
 346 reduction.

347 The PSD operates in the wake of the ship, and hence the size and appear-
 348 ance of the wake field may have a strong influence on the PSD performance.
 349 A quantitative measurement of the wake is troublesome for self-propulsion
 350 cases due to the disturbance and induced flow by the propeller. To obtain
 351 a measure correlating with the size of the wake field, the fact that the wake
 352 is dependent on the thickness of the boundary layers along the hull is used.
 353 The boundary layer thickness is in turn related to the wall shear stress, which
 354 correlates with the friction velocity, u^* ,

$$u^* = \sqrt{\frac{\tau_w}{\rho}} = \frac{y^+ \nu}{y}, \quad (1)$$

355 where τ_w is the wall shear stress, ρ the density, ν is the kinematic viscos-
 356 ity and y the height of the near wall cell. The friction velocity naturally
 357 varies along the hull, so instead a simplified measure is applied to obtain
 358 comparable values between the submissions. The indication of the friction
 359 velocity magnitude is here based on average y^+ on the hull below water sur-
 360 face in combination with the near wall cell height at mid-ship. The predicted
 361 power difference is plotted against this variable representing the wall shear
 362 stress in Figure 11, the submission applying wall resolved grids is not in-
 363 cluded. As expected, a higher wall shear stress is generally noted for the
 364 submissions including a hull roughness model, but a large variation is noted.

365 One participant has investigated the influence of hull roughness through the
366 use of the same CFD setup, applying a simplified propeller model, with and
367 without a hull roughness model. These results are indicated in Figure 11.
368 The difference between those two predictions is similar to the vague trend
369 noted amongst the other predictions: an increased wall shear stress implies
370 a better performance of the PSD. There are two outliers to this vague trend:
371 the first one predicting a power difference of above 3% is based on a sim-
372 plified propeller model; the second one does not show a benefit with the
373 PSD installed, despite in general a very high wall shear stress. The reasons
374 behind the second prediction is not fully understood, it is based on sliding
375 mesh interfaces and applies the ordinary SST $k - \omega$ model. There is further
376 scatter in the correlation between the indication of the friction velocity and
377 the power difference, which may be attributed to for instance propeller mod-
378 elling approach. However, despite this and the relatively few computations
379 compared, the results indicate a dependency between the wake field and the
380 PSD performance. The result is not unexpected, considering both general
381 model-scale test experience as well as the prediction for this configuration
382 based on operation in the larger model-scale wake of 4.4% power reduction.
383 It highlights the importance of an accurate and relevant wake field prediction
384 for ship-scale CFD, which puts the light on the uncertainties related to hull
385 roughness modelling.

386 The dependency of the power reduction prediction on other variables,
387 including free surface modelling approach and grid resolution does not show
388 any trends based on this limited set of predictions with large variations in
389 CFD setups between the participants. Further, as indicated in Figure 6 the

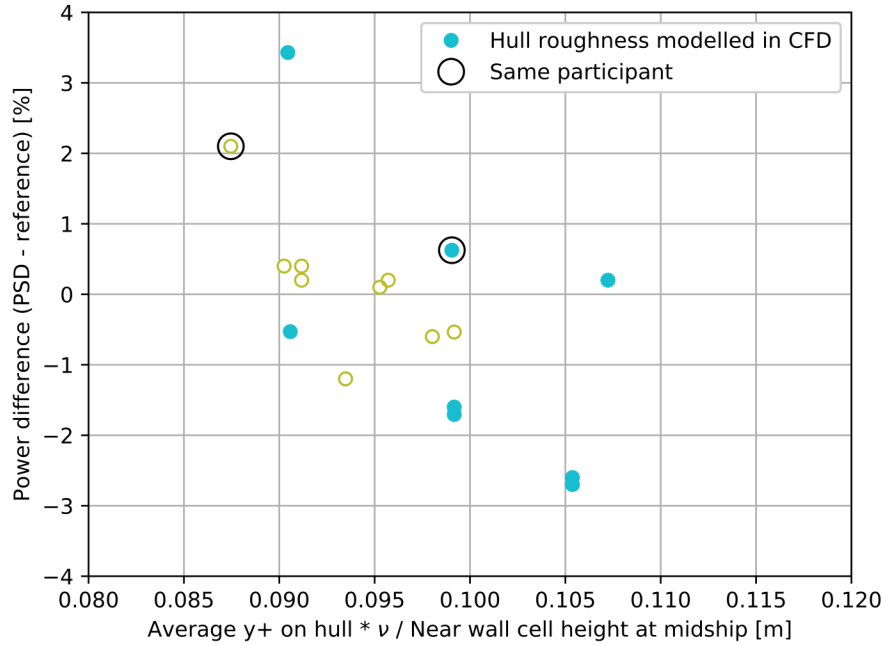


Figure 11: Power difference against an indication of the friction velocity (u^*) magnitude.

390 predicted power difference does not seem to have any notable correlation with
 391 the absolute power within the span predicted within this study. This implies
 392 that factors influencing the load to a minor extent, such as accounting for
 393 hull roughness only through its additional resistance or accounting for the
 394 air resistance, play a minor role.

395 6. Comments on Prediction Uncertainties

396 The predicted power gain as described in Section 5 is around zero, the
 397 average for the whole set of predictions 0.4% with a standard deviation of
 398 1.6%-units and for the subset only including the predictions applying sliding

399 mesh interfaces on average 1.2% with a standard deviation of 1.0%-units.

400 Each participant was asked to list all uncertainties related to their predic-
401 tions, and assess the influence of this method/assumption on the final energy
402 saving prediction. 10 out of the 22 submissions include such an uncertainty
403 description. Six of the 10 estimates included uncertainties rated as "high",
404 which was predefined as factors which may influence the power saving pre-
405 diction with $\pm 50\%$, defined as that an power reduction of 1%, might as well
406 represent a 1.5 or 0.5% power reduction. The most common uncertainties
407 rated as "high" were propeller modelling approach and spatial discretization.
408 Based on the variations in predicted power reduction, it also seems like the
409 propeller modelling approach may have a large influence on the results. In-
410 teresting to note is that only four submissions brought up the hull roughness
411 modelling as an uncertainty in their modelling at all, of which two assessed it
412 as a moderate uncertainty and two as a low uncertainty. Considering the de-
413 pendency of the power reduction against the wall shear stress, as illustrated
414 in Figure 11, it seems like a majority of the participants underestimate the
415 importance of the wake field prediction and hull roughness modelling. An-
416 other important remark based on the self-assessed uncertainties is that they
417 in general are low in relation to the standard deviation of the predicted power
418 gains.

419 **7. Conclusions**

420 The 10 participants, including 13 separate organizations, in this ship-
421 scale CFD benchmark study conducted in total 22 different predictions of
422 the power reduction through a PSD installation on KVLCC2. The predicted

423 power reduction is varying around zero, on average 0.4%, with a standard
424 deviation of 1.6%-units, if not considering two predictions based on model-
425 scale CFD and two outliers associated with large uncertainties in the results.
426 A majority of the predictions were obtained using commercially useful meth-
427 ods in terms of cost and delivery time, claiming approximately one week for
428 delivery of prognosis, counted from that the complete geometry and list of
429 operating conditions are obtained.

430 From this comparative study, two factors could be observed to influence
431 the predicted power reduction: the propeller modelling approach as well as
432 the boundary layer/wake field prediction. A geometrically resolved propeller
433 model applying sliding mesh interfaces to simulate the propeller motion is,
434 based on the set of submitted results, in general predicting a higher power
435 reduction with the PSD compared to simplified propeller models. The reason
436 behind this observation is not fully understood, but a possible theory may
437 be that the representativeness of the simplified propeller models is lower at
438 off-design conditions and although the propeller operates at the design point,
439 each individual blade will experience a large variety of operating conditions
440 during one revolution. This shows on the importance of applying the same
441 propeller models when comparing alternatives, and also indicates that com-
442 parisons of ESD alternatives with different working principles may be sensi-
443 tive to the propeller modelling approach. An indication of the boundary layer
444 thickness is in this study obtained indirectly through a measure indicating
445 the relative magnitude of the wall shear stress. This variable shows a vague
446 correlation towards the predicted power reduction, where thicker boundary
447 layers gives higher power savings. Factors influencing the wall shear stress

448 are mainly the hull roughness modelling, but also the turbulence model and
449 its near wall modelling.

450 Hull roughness modelling is a modelling aspect currently associated with
451 high levels of uncertainty, due to scarce amount of data covering both flow
452 measurements and detailed hull surface characterizations on ships. In this
453 study nine out of 22 predictions account for the hull roughness in the CFD
454 setup, using a variety of roughness functions and equivalent sand grain rough-
455 ness heights, with a varying impact on the wall shear stress. This is also an
456 aspect of highest importance for the ship-owners as it indicates that the
457 PSD performance is dependent on the fouling rate and general hull surface
458 condition of the ship. Further, it indicates that the PSD performance most
459 probably varies in the period between two dry dockings, as a function of the
460 wake field alteration.

461 Factors with no obvious correlation to the predicted power reduction
462 based on the set of submitted results include free surface modelling ap-
463 proach, grid resolution and propeller loading (i.e. absolute power). It is
464 however very important to keep in mind that this observation is based on a
465 limited set of predictions with large variations in CFD setups between the
466 participants. The absolute power seems not to correlate with the predicted
467 power reduction. This implies that factors influencing the load negligibly,
468 such as accounting for hull roughness only through an additional resistance
469 or air resistance, play a minor role. The fact that several factors do not
470 show a correlation with the predicted power reduction raises the question of
471 the necessity to include these in the CFD-model. To exclude modelling of
472 for instance the free-surface, super-structure drag or hull motions, is always

473 associated with a risk since it may have an influence on the results for the
474 specific case studied. On the other hand, all introduced modelling may also
475 imply additional uncertainties when comparing two similar cases. This is
476 clearly illustrated by one of the outliers in this study, where the results are
477 heavily influenced by differences in predicted sinkage and trim.

478 For future studies, to increase the general maturity and trustworthiness of
479 ship-scale CFD, the importance of flow-field measurements in combination
480 with detailed hull surface characterizations on ships is emphasized. This
481 could hopefully facilitate a development within the field of hull roughness
482 modelling for ship-scale CFD which is required. The influence of the selection
483 between alternative propeller models still needs further work, particularly
484 with their applicability to a wide range of operational conditions. While it
485 is attractive, especially for design optimization to use a lower computational
486 cost approach, this cannot be at the expense of failing to resolve the physics
487 of the performance gain associated with an ESD.

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